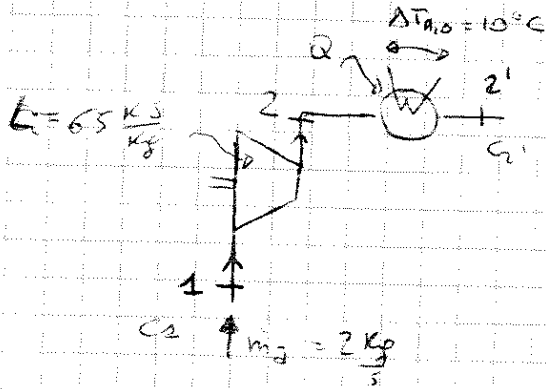


①  $m_a = 2 \text{ kg/s}$   $T_1 = 288 \text{ K}$

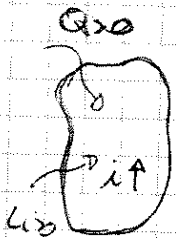
comprensibile e poi irreversibile fino a  $T_2 = 300 \text{ K}$



Sistema aperto da studiare con equazione bilancio

$$Q + L_i = \Delta i^* + \Delta E_c + \Delta E_g + \Delta E_p$$

$\Delta i^* < \Delta i$   
 $\Delta i_{ch}$  se si hanno reazioni chimiche nei sistemi.



In questo caso non abbiamo reazioni chimiche

$$\Delta i_{ch} = 0 \quad \Delta i^* = \Delta i$$

$\Delta E_g$  è trasmissibile perché fluido è aria!!

$\Delta E_p = 0$   
 $\Delta E_p$  è trasmissibile perché mi interessa solo  $\Delta E_p$  tra 1 e 2'

$\Delta E_c$  è trasmissibile perché non è il mio scopo rendere la velocità dell'aria. Suppongo quindi  $C_1 = C_2$

$$Q + L_i = \Delta i \Big|_1^{2'}$$

il lavoro è espresso a grande scala kg di patate che perdono

$$\Delta i = c_p \Delta T = c_p (T_2 - T_1)$$

$$Q = \Delta i - L_i = \left[ 1,005 (300 - 288) - 65 \right] \frac{\text{kJ}}{\text{kg}} = \boxed{-52,8 \frac{\text{kJ}}{\text{kg}}}$$

$$c_p = \frac{k}{k-1} R = 1,005 \frac{\text{kJ}}{\text{kg K}} \quad k = 1,4 \quad R = 287 \frac{\text{J}}{\text{kg K}}$$

Per valutare  $T_2$  devo cambiare volume di controllo

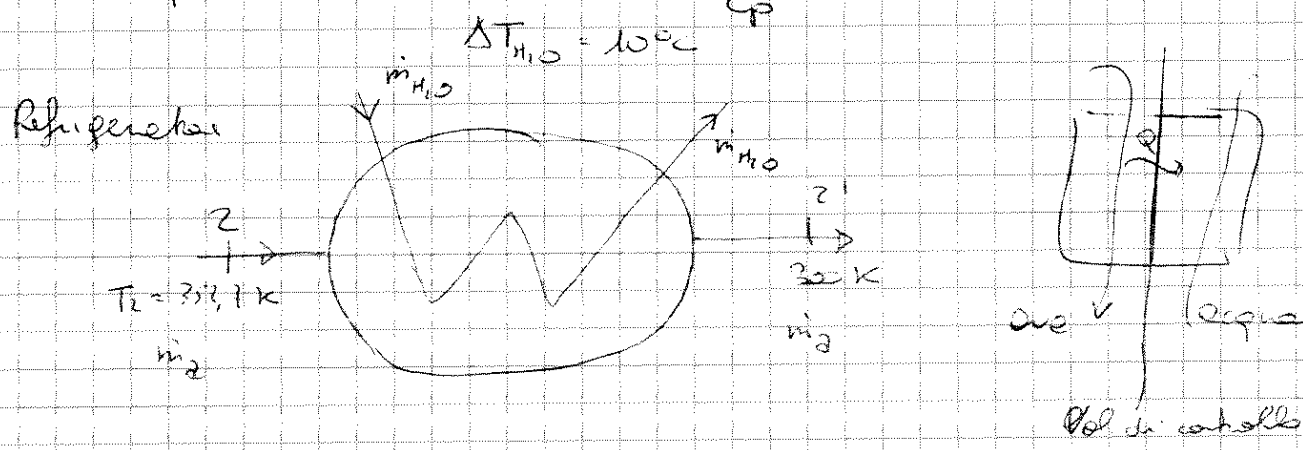
$$\frac{2}{1} \quad Q + L_i = \Delta i \Big|_1^2$$

Hp. la macchina è adiabatica ( $Q = 0$ )

$$L_i = \Delta i \Big|_1^2$$

$$\Delta i \Big|_1^2 = c_p (T_2 - T_1)$$

$$L_i = c_p (T_c - T_2) \quad T_L = \frac{L_i + c_p T_2}{c_p} = 352,7 \text{ K}$$



Q calcolato precedentemente è riferito all'acqua

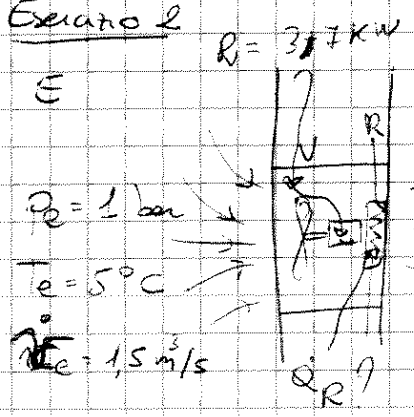
$$\dot{L}_i + \dot{Q}_{H_2O} = \dot{L}_i + \dot{Q}_{H_2O} = 0$$

$$\dot{m}_a \dot{Q}_a = \dot{m}_{H_2O} \dot{Q}_{H_2O}$$

$$\dot{m}_a \dot{Q}_a = \Delta i_{H_2O} \dot{m}_{H_2O}$$

$$c_{p,H_2O} \Delta T \dot{m}_{H_2O} = \dot{m}_a \dot{Q}_a \Rightarrow \dot{m}_{H_2O} = \frac{\dot{m}_a |\dot{Q}_a|}{c_{p,H_2O} \Delta T} = 253 \frac{\text{kg}}{\text{s}}$$

Esercizio 2



A  
 $p_A = p_E = 1 \text{ bar}$   
 $T_A = 35^\circ \text{C}$   
 $v_A = 0$

$\eta_M = 0,87$   
 $R_{spec} = 287 \frac{\text{J}}{\text{kg K}}$   
 $c_p = 1005 \frac{\text{J}}{\text{kg K}}$

$$Q + L_i = \Delta i \Big|_E^A + (\Delta E_c, \Delta E_g, \Delta E_{cg})$$

$$\dot{Q}_R = \dot{L}_i \Big|_E^A - L_i \quad \Delta i \Big|_E^A = c_{p,a} (\Delta T) = c_{p,a} (T_A - T_E)$$

$$P_i = \eta_M P_e = 0,87 \cdot 37 = 3,219 \text{ kW}$$

$$L_i = \frac{P_i}{\dot{m}_a}$$

$$\dot{m}_a = p_e \dot{V}_e \quad \frac{p}{\rho} = RT \quad \rho = \frac{p_e}{RT_e} = \frac{10^5}{297 \cdot 273,15} = 1,253 \frac{\text{kg}}{\text{m}^3}$$

$$L_i = \frac{P_i}{\dot{m}_a} = \frac{3,528}{1,88} = 1,878 \frac{\text{kJ}}{\text{kg}}$$

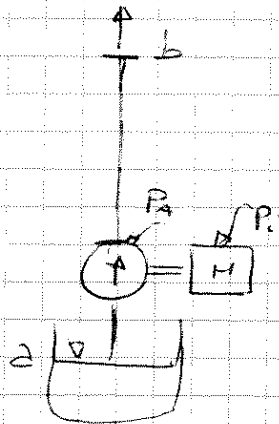
$$Q_R = c_p (T_A - T_C) - L_i = 1,005 (35 - 5) - 1,878 = 7,261 \frac{\text{kJ}}{\text{kg}}$$

$$\dot{Q}_R = \dot{m}_a Q_R = 1,88 \cdot 7,261 = 13,65 \text{ kW}$$

## Esercitazione 2

21/03/2012

①



$z_b - z_A = 20 \text{ m}$       $d = 10 \text{ cm}$  diametro condotto

$$\eta_M = 0,97 \quad \dot{Q}_b = 2 \frac{\text{m}}{\text{s}}$$

il sistema è aperto

$$Q + L_i = \Delta E_c + \Delta E_p + \Delta E_f + \Delta E_{\text{irr}} = 0$$

$$\Delta E_c = g(z_b - z_A) \quad \Delta E_p = \frac{v_b^2 - v_A^2}{2} \quad v_A = 0$$

$$L_i = \int_2^b v dp + L_w + \Delta E_c + \Delta E_p \quad \int_2^b v dp = \int_2^b \frac{1}{\rho} dp = \frac{1}{\rho} (P_b - P_A) = 0$$

perché  $P_A = P_b = P_{atm}$

$$L_i = L_w + \Delta E_c + \Delta E_p = L_w + \frac{v_b^2}{2} + g \Delta z$$

$$B = \frac{P_i}{\eta_m} = \frac{\dot{m} L_i}{\eta_m}$$

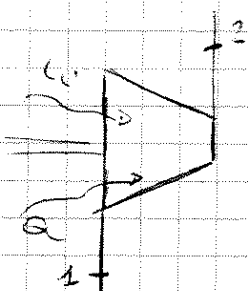
$$\dot{m} = \rho_b \cdot A_b \cdot v_b = \rho \pi \frac{d^2}{4} v_b = 15,7 \frac{\text{kg}}{\text{s}}$$

Supponiamo che  $L_w = 0$       $L_i = 197,2 \frac{\text{J}}{\text{kg}} = \Delta P_b = 3,21 \text{ kW}$

Supponiamo  $L_w = 15\% L_i$       $L_i (0,85) = \frac{v_b^2}{2} + g \Delta z = \Delta P_b = 233,2 \frac{\text{J}}{\text{kg}}$

$$P_b = 3,77 \text{ kW}$$

②



$$T_1 = 17^\circ \text{C}$$

$$P_2 = 200 \text{ kPa}$$

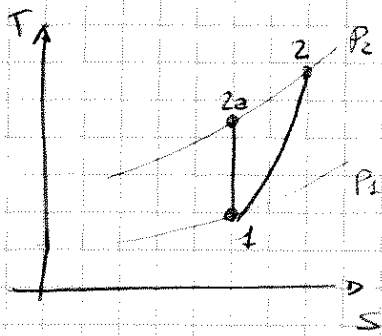
$$P_1 = 100 \text{ kPa}$$

$$\Delta E_c = 0$$

$$K = 1,4$$

$$c_p = 1004 \frac{\text{J}}{\text{kg K}}$$

2) ipotesi di trasformazione adiabatica reversibile



$$\frac{dT}{ds} = \frac{T}{c_p}$$

$$Q - L_i = \Delta i + \Delta \phi_c + \Delta \phi_f + \Delta \phi_g$$

$$Q - L_i = c_p \Delta T$$

~~poli trof. adiabatica~~

Curve 1-2 è politropica con  $m = 1,55$

~~è in assenza della~~

$$\frac{T_{2a}}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}} \quad T_{2a} = T_1 \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}} = 80,5^\circ\text{C}$$

Poiché supponiamo trof. adiabatica  $Q=0$  e quindi

$$\alpha_i = c_p \Delta T = 63,8 \frac{\text{KJ}}{\text{kg}}$$

b) Supponiamo trof. adiabatica con attutiti ( $m = 1,55$ )

$$Q + \alpha_i = c_p \Delta T$$

$\underset{=0}{Q}$   
adib.

$$T_{1b} = T_1 \left(\frac{P_2}{P_1}\right)^{\frac{m-1}{m}} = 340,9 \text{ K} \Rightarrow \alpha_i = c_p \Delta T = 81,3 \frac{\text{KJ}}{\text{kg}}$$

$$\alpha_i = \int_1^{2b} v dp + \alpha_w + \Delta \phi_c \Rightarrow \int_1^{2b} v dp = \frac{m}{m-1} R (T_{2b} - T_1)$$

$$\alpha_w = \alpha_i = \frac{m}{m-1} R (T_{2b} - T_1) = 15,3 \frac{\text{KJ}}{\text{kg}}$$

$$\alpha_i = \alpha_{i,s} + \alpha_w + L_{CR} \Rightarrow L_{CR} = \alpha_i - \alpha_{i,s} - \alpha_w = 1,6 \frac{\text{KJ}}{\text{kg}}$$

$$\eta_{15} = \frac{\alpha_{i,s}}{\alpha_i} = 0,785$$

$$\eta_y = \frac{\alpha_i - \alpha_w}{\alpha_i} = 0,804$$

$\eta_{15} < \eta_y$  compressore

$\eta_y < \eta_{15}$  turbina

c) refrigerazione senza attutiti ( $m = 1,28$ )

In questo caso  $s < 0$   $q < 0$  e  $L_w = 0$

$$Q + L_i = c_p (T_{2c} - T_1)$$

$$T_{2c} = T_1 \left(\frac{P_2}{P_1}\right)^{\frac{m-1}{m}} = 64,5^\circ\text{C} \quad \alpha_i = \int_1^{2c} v dp = \frac{m}{m-1} R (T_{2c} - T_1) = 63,3 \frac{\text{KJ}}{\text{kg}}$$

$$Q = c_p (T_{2c} - T_1) - L_i = -24,6 \frac{\text{KJ}}{\text{kg}}$$

d) Raffreddamento ~~isotermo~~ <sup>isotermo</sup> senza effetti:

$$Q + L_i = c_p (T_{2d} - T_1) \quad T_{2d} - T_1 = 0$$

$$Q = -L_i$$

$$L_i = \int_1^{2d} v dp = RT_1 \int_1^{2d} \frac{dp}{p} = RT_1 \ln\left(\frac{P_2}{P_1}\right) = 51,7 \frac{kJ}{kg}$$

$$Q = -51,7 \frac{kJ}{kg}$$

e) raffreddamento isotermo con effetti:  $L_w = 15,9 \frac{kJ}{kg}$

$$T_{2d} = T_{1e} = T_1$$

$$Q = -L_i = -\int_1^{2e} v dp + L_w = -73,6 \frac{kJ}{kg}$$

### Esercizio 3

1) Ugello convergente, espansione isentropica

2)

$P_0 = 2 \text{ bar}$        $P_1$  pressione nell'ambiente e nell'ugello  
 $m = 3 \frac{kg}{s}$  aria  
 $R = 287 \frac{J}{kg \cdot K}$        $\kappa = 1,4$   
 $T_0 = 150^\circ C$   
 $v_0 = 0$

$$\underbrace{Q}_{=0} + \underbrace{L_i}_{=0} = \Delta i + \Delta E_c + \underbrace{\Delta \left(\frac{v^2}{2}\right)}_{\approx 0} + \underbrace{\Delta \left(\frac{v^2}{2}\right)}_{\approx 0}$$

supponendo  
sia adiabatico

$$0 = \Delta i + \Delta E_c$$

Dobbiamo risolvere per avere in condizioni critiche o meno.

$$\frac{P_{u,c}}{P_0} = \left(\frac{T_{u,c}}{T_0}\right)^{\frac{\kappa}{\kappa-1}} = \left(\frac{2}{\kappa+1}\right)^{\frac{\kappa}{\kappa-1}} = 0,528$$

$P_0$  e  $T_0$  sono le condizioni totali.

$$\frac{T_{u,c}}{T_0} = \frac{2}{\kappa+1} = \frac{2}{1,4+1}$$

$$P_0 = P_0 \left(\frac{T_0}{T_0}\right)^{\frac{\kappa}{\kappa-1}} = P_0$$

$$T_0 = T_0 + \frac{c_0^2}{2 c_p} = T_0$$

temperatura  
x fluido è fissa

$$c_0 \approx 0$$

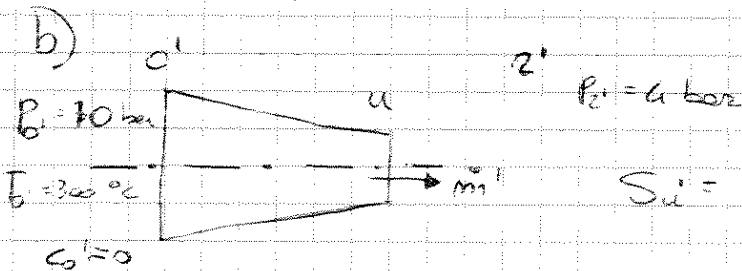
$$T_{u,c} = \frac{2}{\kappa+1} \cdot T_0$$

$$\frac{P_2}{P_0} = \frac{2}{5} = 0,4 < 0,528 \quad \text{nono in condizioni critiche!!}$$

$$P_{u, T_{u,cr}} = \frac{2}{k+1} P_0 = \frac{2}{2,4} (150 + 273) = 352,5 \text{ K} = 79,5^\circ \text{C}$$

$$c_{u,cr} = \sqrt{k R T_{u,cr}} = \sqrt{1,4 \cdot 287 \cdot 352,5} = 376 \text{ m/s}$$

$$\dot{m} = \rho_{u,cr} \cdot c_{u,cr} \cdot S_u \quad \rho_{u,cr} = \frac{P_{u,cr}}{R T_{u,cr}}$$



$$S_{u1} = S_u$$

$$\frac{P_{u,cr}}{P_0} = f(k) = 0,528$$

$$\frac{P_2}{P_0} = \frac{4}{10} = 0,4 < 0,528 \quad \text{condizioni critiche}$$

$$\dot{m}' = \rho_{u,cr}' \cdot c_{u,cr}' \cdot S_u$$

$$c_{u,cr}' = \sqrt{k R T_{u,cr}'} =$$

$$T_{u,cr}' = \frac{T_0' \cdot 2}{k+1}$$

$$\rho_{u,cr}' = \frac{P_{u,cr}'}{R T_{u,cr}'}$$

$$\dot{m}'_u = \frac{P_{u,cr}'}{R T_{u,cr}'} \cdot \sqrt{k R T_{u,cr}'} \cdot S_u = \frac{P_{u,cr}' \cdot \sqrt{k}}{\sqrt{R T_{u,cr}'}} \cdot S_u =$$

$$= \frac{P_0' \cdot \left(\frac{2}{k+1}\right)^{\frac{k}{k-1}}}{\sqrt{R \cdot \frac{T_0' \cdot 2}{k+1}}} \cdot \sqrt{k} \cdot S_u = \frac{P_0'}{\sqrt{T_0'}} \cdot S_u \cdot \underbrace{f(k, R)}_{\text{tipo di gas}}$$

finché il getto è sottile le sue pareti dipendono solo dal rapporto  $\frac{P_0'}{\sqrt{T_0'}}$

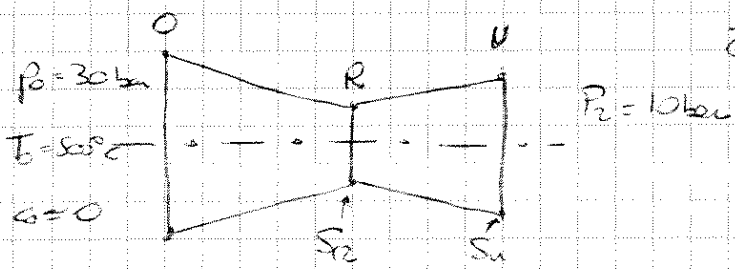
$$\dot{m}'_u = \frac{P_0'}{\sqrt{T_0'}} S_u f(k, R)$$

$$\dot{m}_u = \frac{P_0}{\sqrt{T_0}} S_u f(k, R)$$

$$\dot{m}'_u = \dot{m}_u \cdot \frac{P_0'}{P_0} \sqrt{\frac{T_0}{T_0'}}$$

$$\dot{m}'_u = 3 \cdot \frac{10}{5} \sqrt{\frac{150+273}{300+273}} = 5,16 \frac{\text{kg}}{\text{s}}$$

2) ugelli convergente-divergenti distribuiti da una turbina a vapore - Espansione isentropica

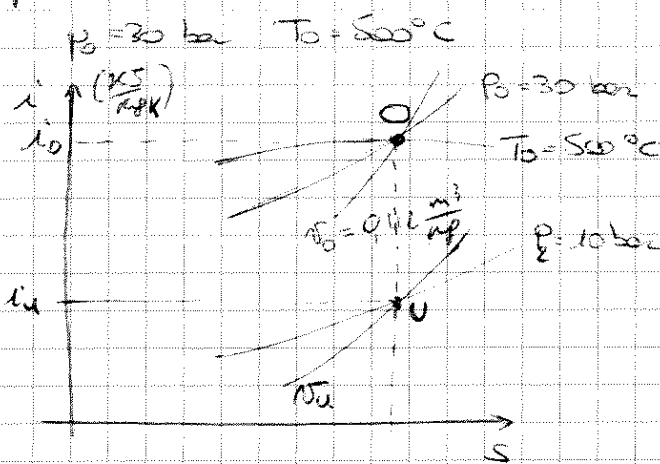


fluido vapore d'acqua

$$\left. \begin{aligned} p_0 &= p_0 \\ T_0 &= T_0 \end{aligned} \right\} \rightarrow G = 0$$

$$\frac{p_R}{p_0} = \frac{10}{30} = 0,33$$

ipotesi  $p_u = p_2$



$$v_0 = \frac{3660}{3660} \frac{\text{kg}}{\text{kg}}$$

$$v_u = 3120 \frac{\text{kg}}{\text{kg}}$$

$$v_u = \frac{1}{\rho_u} = 0,320 \frac{\text{m}^3}{\text{kg}}$$

$$0 = \Delta i + \Delta \bar{e}$$

$$0 = (i_u - i_0) + \frac{c_u^2 - c_0^2}{2}$$

$$c_u = \sqrt{(i_0 - i_u) \cdot 2} = 826,6 \text{ m/s}$$

$$\dot{m} = \rho_u c_u S_u$$

$$S_u = \frac{\dot{m}}{\rho_u c_u} = \frac{\dot{m} v_u}{c_u} = 0,0011035 \text{ m}^2$$

$$= 11,035 \text{ cm}^2$$

So che R è tra O e U lungo l'espansione isentropica.

Possiamo fare della approx per valutare un  $k_v$ .

$$p_0 v_0^{k_v} = p_u v_u^{k_v}$$

$k_v$  è un  $k$  equivalente <sup>velocità</sup> limitatamente all'espansione O-U

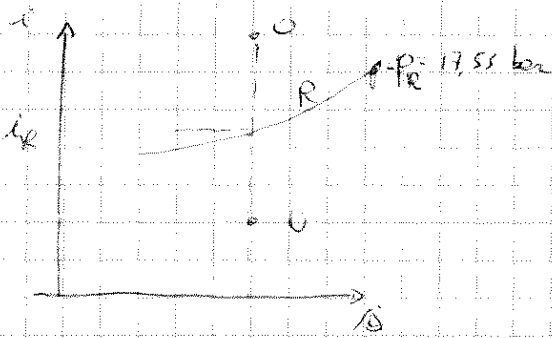
$$k_v = \frac{\ln\left(\frac{p_0}{p_u}\right)}{\ln\left(\frac{v_0}{v_u}\right)} = \frac{\ln\left(\frac{30}{10}\right)}{\ln\left(\frac{0,360}{0,12}\right)} \approx 1,30$$

$$\frac{p_{R,1}}{p_0} = \left(\frac{2}{2,211}\right)^{\frac{1}{1,3}} = 0,585 >> 0,585$$

nelto prob. cioè in condizione critica

Quindi in R viene rimanente la pressione critica

$$P_R = P_{R,c} = 0,585 \cdot P_0 = 17,55 \text{ bar}$$



$$i_2 = 3280 \frac{\text{kJ}}{\text{kg}}$$

$$c_R = \sqrt{2(i_0 - i_2)} = 600 \text{ m/s}$$

$$\text{tra } O \text{ e } R \quad c = \Delta c_1 + \Delta c_2$$

L'approx sta nel calcolo di  $i_2$   
 Se  $i_2$  è giusta allora  $c_R$  è corretta!!

Se invece risulta  $c_R = \sqrt{K_v R_v T_{R,c}} = \sqrt{1,3 \cdot 17,55 \cdot 10^3 \cdot 0,178} = 638,6 \text{ m/s}$

$$P_{R,c} = P_{R,c}$$

sbagliato perché abbiamo considerato il vapore come un gas ideale!!!

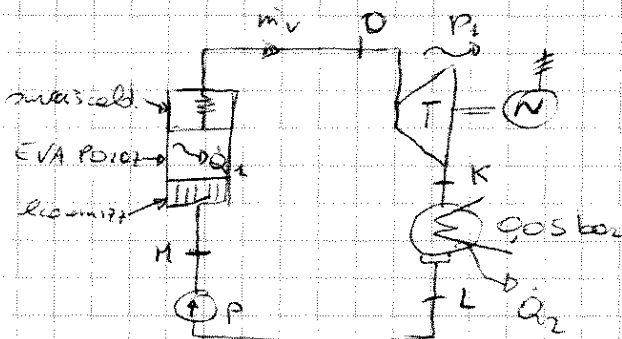
$$S_R = \frac{3,5 \frac{\text{m}^3}{\text{s}}}{P_R c_R} \approx 10,6 \text{ cm}^2$$

$$P_i = P_u$$

Lipoten è corretta perché il vapore espande curve in fondo all'ipello in condizioni di espansione

## ESERCITAZIONE 4

1) Impianto a vapore non regenerativo e senza surriscaldamento



$$P_0 = 100 \text{ bar} \quad T_c = 550 \text{ °C}$$

$$P_c = P_k = 9,05 \text{ bar}$$

$$\eta_{is} = 0,8$$

$$\eta_0 = 0,88$$

$$\eta_u = ?$$

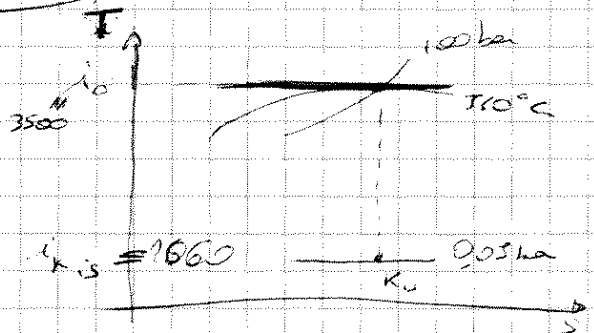
$$\eta_u = \frac{L_u}{Q_1} = \eta_0 \frac{L_i}{Q_1} = \eta_0 \eta_i$$

$$L_i = L_c - \frac{L_p}{\eta_p} \approx L_c = i_0 - i_k$$

$$i_{k,is} = 1860$$

$$A_k = 0,03 \text{ m}^2$$

trascurabile rispetto a  $L_c$

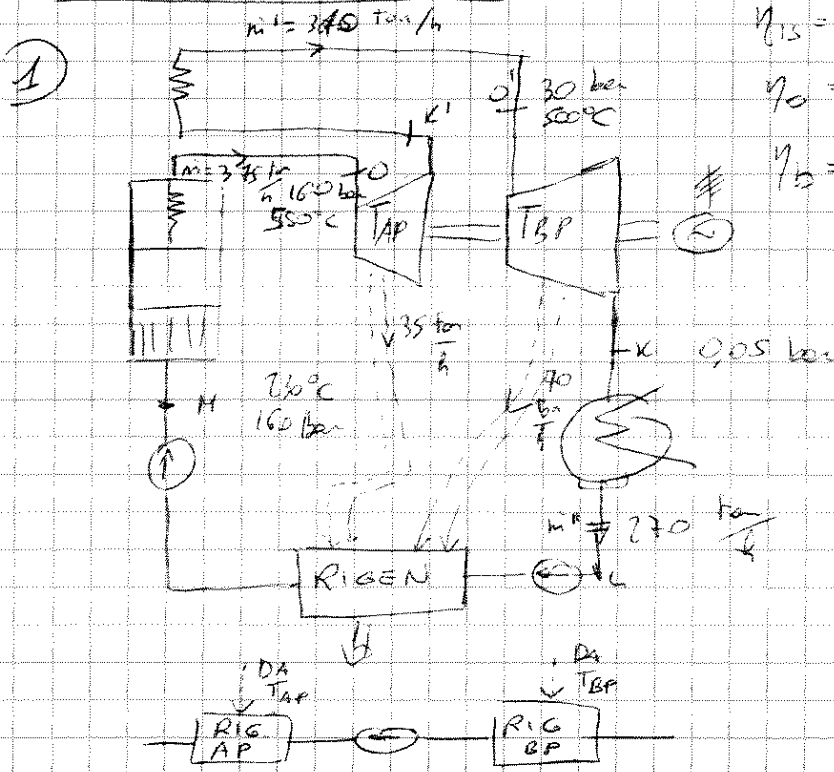








# ESERCITAZIONE 5



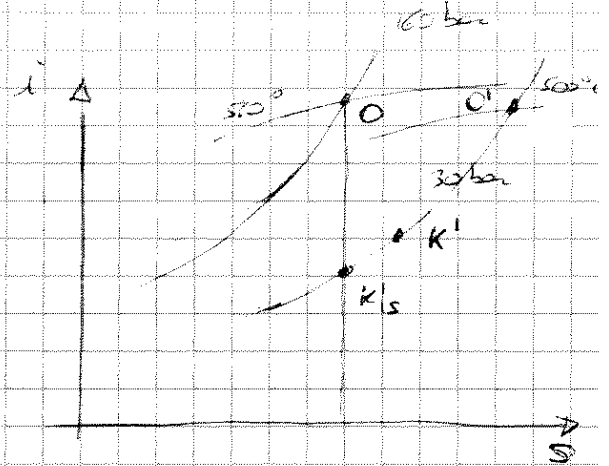
$$\eta_{13} = 0,8$$

$$\eta_0 = 0,97$$

$$\eta_2 = 0,84$$

$$P_u = \eta_0 P_i$$

$$P_i = \dot{Q}_1 - \dot{Q}_2$$



$$\dot{Q}_1 = \dot{m} (i_0 - i_M) + \dot{m}' (i_{k'} - i_{k15})$$

$$\dot{Q}_2 = \dot{m}'' (i_k - i_L)$$

$$i_0 = 3438 \frac{\text{KJ}}{\text{kg}}$$

$$i_{k15} = 2960 \frac{\text{KJ}}{\text{kg}}$$

$$i_{k'} = 3457 \frac{\text{KJ}}{\text{kg}}$$

$$i_M = \left( \frac{c_{p,i}}{270^\circ\text{C}} \right) = 2960 \frac{\text{KJ}}{\text{kg}}$$

$$i_{k'} = i_0 - \eta_{13} (i_0 - i_{k15}) = 3438 - 0,8 (3438 - 2960) = 3055,6 \frac{\text{KJ}}{\text{kg}}$$

$$\dot{Q}_1 = \left[ 345 \frac{\text{ton}}{\text{h}} (3438 - 2960) + 340 \frac{\text{ton}}{\text{h}} (3457 - 3055,6) \right] \frac{1}{3,6} = 292,9 \text{ MW}$$

$$i_k = i_{k'} - \eta_B (i_{k'} - i_{k15}) = 2456$$

$$i_L = \left( \frac{c_{p,i}}{270^\circ\text{C}} \right) = 137,7 \frac{\text{KJ}}{\text{kg}}$$

$$\dot{Q}_2 = \frac{1}{3,6} (270 (2456 - 137,7)) = 173,9 \text{ MW}$$

$$P_u = \eta_0 (\dot{Q}_1 - \dot{Q}_2) = 0,97 (292,9 - 173,9) = 115,4 \text{ MW}$$



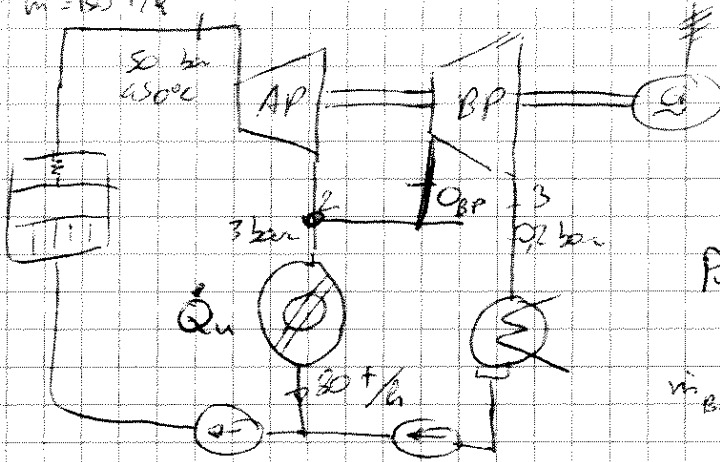
$$\dot{m}_b = \frac{P_u}{\eta_b \eta_o \eta_s \eta_{15}} = \frac{7,25 \frac{\text{kW}}{\text{s}}}{0,8 \cdot 0,9 \cdot 0,9 \cdot 0,9} = 11,03 \frac{\text{kg}}{\text{s}}$$

$$\dot{m}_b = \frac{P_u}{\eta_b \eta_o \eta_{15} \eta_{10}} = \dots$$

## ESERCITAZIONE 6

1)

2)  $\dot{m} = 150 \text{ t/h OAP}$

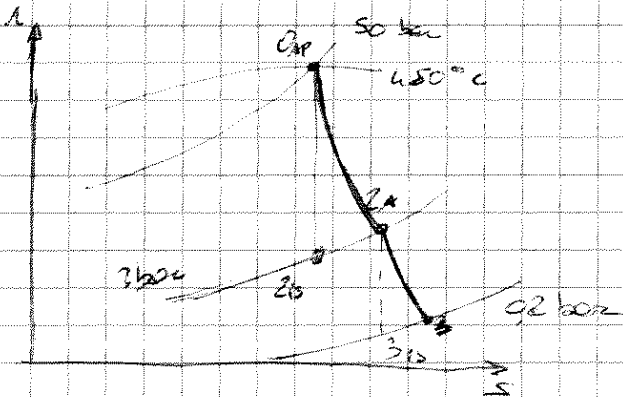


$$\eta_o = 0,87$$

$$\eta_{AP} = \eta_{BP} = 0,8$$

$$P_u = \eta_o P_i = \eta_o [\dot{m}_{AP} L_{iAP} + \dot{m}_{BP} L_{iBP}]$$

$$\dot{m}_{BP} = 150 - 20 = 130 \text{ t/h}$$



$$L_{iAP} = \eta_{AP} (i_{3AP} - i_{4AS}) = 526,1 \frac{\text{kJ}}{\text{kg}}$$

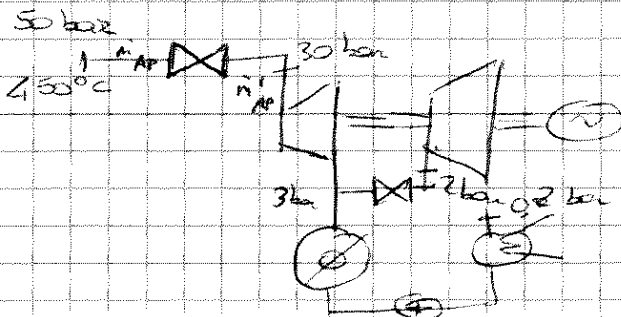
$$i_2 = i_{1BP} - L_{iAP} = 2726 \frac{\text{kJ}}{\text{kg}}$$

$$L_{iBP} = \eta_{BP} (i_{1BP} - i_{315}) = 248,8 \frac{\text{kJ}}{\text{kg}}$$

$$i_3 \approx 2350 \frac{\text{kJ}}{\text{kg}}$$

$$P_u = \eta_o (\dot{m}_{AP} L_{iAP} + \dot{m}_{BP} L_{iBP}) = 27,76 \text{ MW}$$

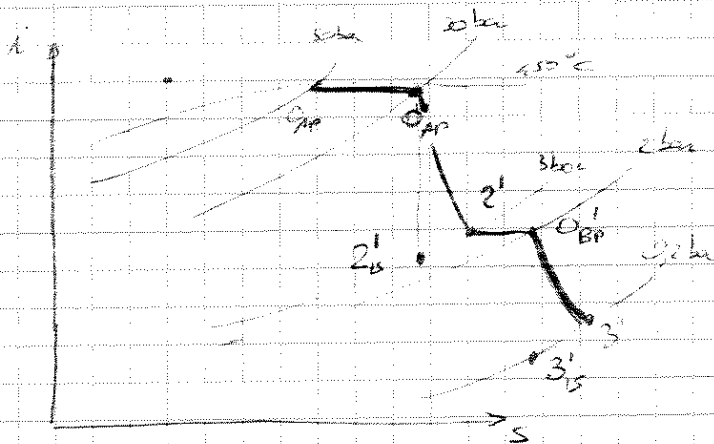
b) 2 velocità di laminazione



$$\dot{m}'_{AP} = \dot{m}'_{AP} \frac{P_{AP}}{P_{BP}} = 150 \frac{30}{50} = 90 \text{ t/h}$$

$$\dot{m}'_{BP} = \dot{m}'_{BP} \frac{P_{BP}}{P_{BP}} = 70 \frac{2}{3} = 46,67 \text{ t/h}$$

$$\dot{m}'_u = 43,33 \text{ t/h}$$



$$L'_{AP} = \eta'_{AP} (\dot{m}'_{AP} i'_{2AP} - \dot{m}'_{AP} i'_{1AP}) = 428 \frac{\text{KS}}{\text{kg}}$$

$$i'_{21} = 2882 \frac{\text{KS}}{\text{kg}}$$

$$L'_{BP} = \eta'_{BP} (\dot{m}'_{BP} i'_{2BP} - \dot{m}'_{BP} i'_{1BP}) = 317,6 \frac{\text{KS}}{\text{kg}}$$

$$i'_{31} = 2564,6 \frac{\text{KS}}{\text{kg}}$$

$$P_u = \eta_0 (\dot{m}'_{AP} L'_{AP} + \dot{m}'_{BP} L'_{BP}) = 14,38 \text{ MW}$$

$$\dot{Q}'_u = \dot{m}'_u \Delta i'_u = \dot{m}'_u (i'_{21} - i'_{31})$$

$$\dot{Q}'_u = \dot{m}'_u \Delta i'_u = \dot{m}'_u (i'_{2AP} - i'_{3'})$$

Nota  $i'_j$  è funzione della temperatura

2) Analisi di uno stadio di turbina a elica

$$\dot{m} = 150 \frac{\text{kg}}{\text{s}}$$

$$P_0 = 80 \text{ bar}$$

$$T_0 = 500^\circ \text{C}$$

$$\alpha_0 = 0$$

$$P_1 = 40 \text{ bar}$$

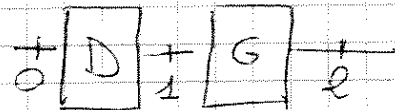
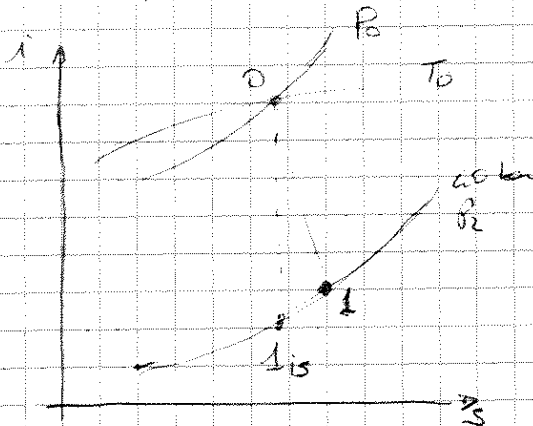
$$\alpha_1 = 30^\circ$$

$$\frac{u}{c} = \frac{1}{2} \cos \alpha_1$$

relazione macchina  $n = 3000 \text{ rpm}$

$$\psi = 0,85$$

$$\phi = 0,8$$



Pochi merchia ad azione

$$P_1 = P_2$$

1° principio di distibutor

$$\dot{Q}'_i = \dot{Q}'_e + \dot{W}'_i + \dot{W}'_e + \dot{Q}'_f + \dot{Q}'_g \quad \Delta i = \Delta i_{0e}$$

Se esp. necessaria

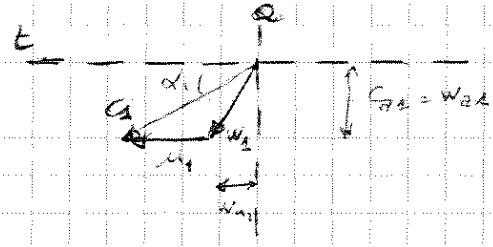
$$0 = A_{115} - C_1 + \frac{C_{115}^2 - C_1^2}{2}$$

$$10^0 = 10 + \frac{C_1^2}{2} \approx 10$$

$$0 = i_{115} - 10^0 + \frac{C_{115}^2}{2}$$

$$C_{115} = \sqrt{2(i_0 - i_{115})} = 66,5 \text{ m/s}$$

$$\psi = \frac{C_{115}}{C_{115}} \Rightarrow C_1 = \psi C_{115} = 612,7 \text{ m/s}$$



$$\frac{u}{C_1} = 0,5 \cos \alpha$$

$$u = \frac{C_1}{2} \cos \alpha = 265,3 \text{ m/s}$$

$$W_1 = \sqrt{W_{12}^2 + W_{11}^2} = 405,3 \text{ m/s}$$

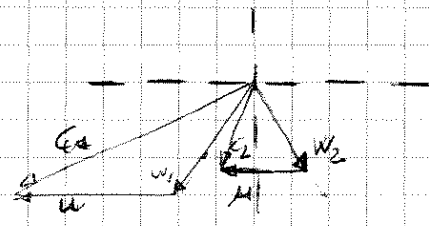
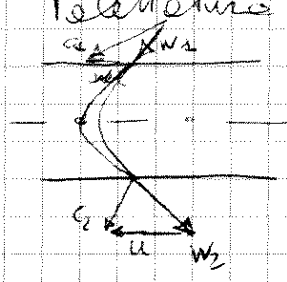
$$W_{12} = C_{115} = C_1 \sin \alpha$$

$$\beta_1 = \arcsin\left(\frac{W_{12}}{W_1}\right) = 49,1^0$$

$$W_{11} = C_{115} - u$$

$$W_{215} = W_1 \quad w_2 = \psi W_{215} = \psi W_{11} = 364,8 \text{ m/s}$$

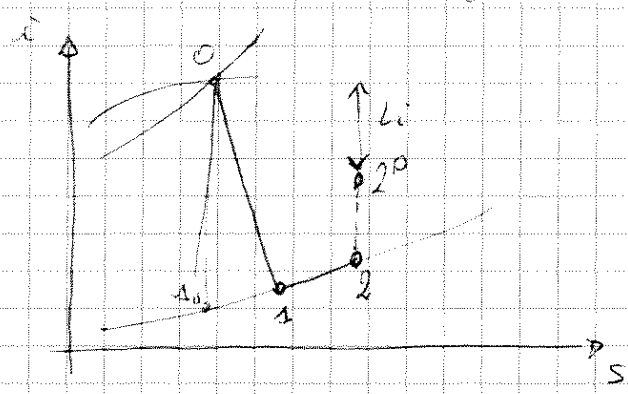
Palettatura simmetrica



$$C_2 = \sqrt{u^2 + C_{115}^2} = 277 \text{ m/s}$$

$$(W_{12} - u)^2 = W_{22}^2$$

$$\alpha_2 = \arcsin\left(\frac{C_{115}}{C_2}\right) = 36,5^0$$



ntao stado (D+G)

$$Q + L_i = R + D + G = 0$$

combene sepe e Li perdei  
vghera vedere caso positivo  
fatto de turbu

$$L_i = -D_1 - D + G =$$

$$= i_0 - i_2 + \frac{C_1^2 - C_2^2}{2} = 10^0 - i_2 - \frac{C_2^2}{2}$$

$$L_i = (C_{115} - C_2) u = 133,7 \frac{\text{m/s}}{\text{kg}}$$

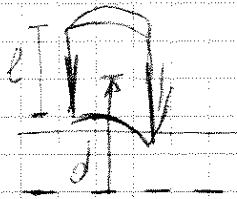
$$i_1 = i_0 + i_0^2 \rightarrow \frac{G^2}{2} = 3212 \frac{\text{KS}}{\text{kg}}$$

$$i_2 = i_1 + \frac{W_{i,2}^2}{2} \left( \frac{1}{\psi^2} - 1 \right) = 3228 \frac{\text{KS}}{\text{kg}}$$

$$\eta_i = \frac{L_i}{i_0 - i_{1,5}} = \frac{133,7 \cdot \frac{1000}{10000}}{\frac{612,7}{7095}} = 0,663$$

LUNGHEZZA SPIGOLO DI INGRESSO

$$l \geq 10 \text{ mm} \quad \frac{l}{d} \geq 0,01$$



$$u = \pi d n = \frac{1}{2} c_{u,2} = 265,3$$

$$d = \frac{265,3}{\pi \frac{3000}{60}} = 1,65 \text{ m}$$

n va espresso in giri/secondo

$$N_1 = 0,040 \frac{\text{m}^3}{\text{kg}}$$

$$\dot{m} = \frac{1}{\sqrt{1}} \pi l d \underbrace{c_1 \rho \alpha_1}_{c_{s1}} \bar{v} \quad \bar{v} \text{ non è stato dato e quindi } \approx 1$$

$$l = \frac{\dot{m} N_1}{\pi d c_1 \rho \alpha_1 \bar{v}} = 6,8 \text{ mm}$$

trovo il grado di perturbation per avere  $l \geq 10 \text{ mm}$

$$\dot{m} = \frac{1}{\sqrt{1}} \pi l d (1 - \epsilon) c_1 \rho \alpha_1 \quad \text{fissiamo } l = 10 \text{ mm}$$

$$\epsilon = 0,32$$

$$\text{ma } \frac{l}{d} < 0,01 \quad \text{vogliamo } l \geq 16,5 \text{ mm}$$

$$\text{ne segue che } 1 - \epsilon = 0,4 \rightarrow \epsilon = 0,6$$

Bisogna perturbation il 60% del perigetto



20/6/12 CSERCITAZIONE

I) Girante turbine aiale reazione (Pacpi)

$$\rho_{20} \left\{ \begin{array}{l} k = \frac{1}{4} \\ R = 20 + \frac{5}{14} k \end{array} \right.$$

$$m = 135$$

$$p_1 = 10 \text{ bar}$$

$$T_1 = 750^\circ \text{C}$$

$$C_1 = 650 \text{ m/s}$$

$$\alpha = 20^\circ$$

$$p_2 = 7 \text{ bar}$$

$$C_2 = p_2 = 28 \text{ m}$$

$$\varphi = 90^\circ$$

$$n = 3000 \text{ rpm}$$

$$u = 300$$

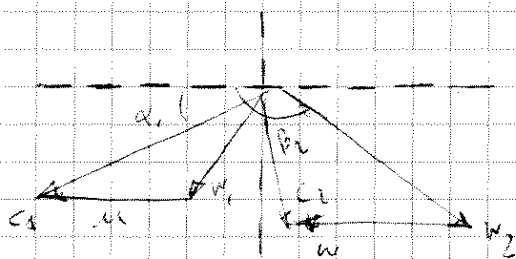
$$P_i = L_i \dot{m} \quad \dot{m} = \rho_1 \int (r dr) C_1 \sin \alpha_1$$

$$\rho_1 = \frac{p_1}{RT_1} = \frac{10 \cdot 10^5}{287 \cdot (273 + 750)} = 3,41 \frac{\text{kg}}{\text{m}^3}$$

$$d = \frac{u}{\pi n} = \frac{300}{\pi \cdot 3000} = 0,0318 \text{ m}$$

$$\dot{m} = 747 \frac{\text{kg}}{\text{s}}$$

$$L_i = q_p (T_2 - T_1) = u (C_{u1} - C_{u2})$$



$$w_1 = \sqrt{(C_1 \cos \alpha_1 + u)^2 + (C_1 \sin \alpha_1)^2} = 180,5 \text{ m/s}$$

$$\beta_1 = \arcsin \left( \frac{C_{1r}}{w_1} \right) = 54,1^\circ$$

Applicando il 1° principio al nodo scelto in figura.

$$0 = \Delta E_c + \Delta E_t + \dots$$

$$0 = \rho p (T_2 - T_1) + \frac{w_1^2 - w_2^2}{2} \Rightarrow w_2 = 669,4 \text{ m/s}$$

$$T_2 = T_1 \left( \frac{p_2}{p_1} \right)^{\frac{\gamma-1}{\gamma}} = 937,6 \text{ K}$$

$$\dot{m} \exp G_2 \times p w_2 \Rightarrow p_1 C_{2s} = p_2 C_{2r}$$

$$w_{2r} C_{2s} \frac{p_1}{p_2} = C_{2s} \left( \frac{p_1}{p_2} \right)^{\frac{1}{\gamma}} = 200,5 \text{ m/s}$$

$$\beta_2 = \arcsin \frac{w_{2r}}{w_2} = 15,7^\circ$$

$$c_2 = \sqrt{(w_2 \cos \beta_2 + u_2)^2 + (w_2 \sin \beta_2)^2} = 735,5 \text{ m/s}$$

$$\alpha_2 = 121,8^\circ$$

$$L_i = u (c_{u1} - c_{u2}) = 164,2 \frac{\text{kJ}}{\text{kg}}$$

$$P_c = m L_i = 172,6 \text{ kW}$$

## 2) Turbocompressore

$$\dot{m} = 25 \text{ kg/s}$$

$$p_1 = 1 \text{ bar}$$

$$T_1 = 15^\circ\text{C}$$

$$p_2 = 3 \text{ bar}$$

$$\eta_{gc} = 0,75$$

$$\eta = 0,9$$

$$\frac{e''}{d''} = 0,4$$

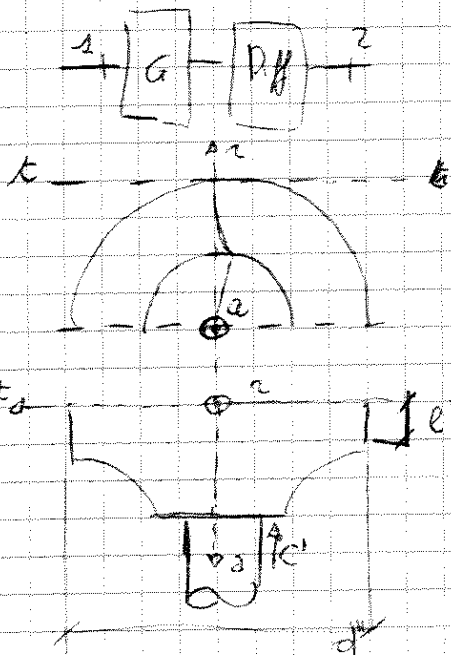
$$\alpha'' = 20^\circ$$

$$m = \text{cost}$$

$$L_i = c_p (T_2 - T_1)$$

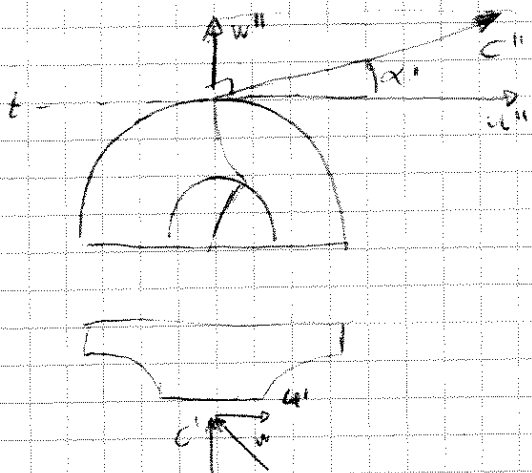
$$T_2 = T_1 \left( \frac{p_2}{p_1} \right)^{\frac{\gamma-1}{\gamma}} = T_1 \left( \frac{p_2}{p_1} \right)^{\frac{1}{\eta_{gc}} \frac{\gamma-1}{\gamma}} = 164,7^\circ\text{C}$$

$$L_i = 134,3 \frac{\text{kJ}}{\text{kg}}$$



$e''$  è altezza delle palette  
in uscita dalle guide

$e'$  è altezza delle palette e  
anello



$$L_i = -u' c_{u1}' + u'' c_{u2}''$$

$$c_{u1}' = 0$$

$$c_{u2}'' = u''$$

$$L_i = (u'')^2$$

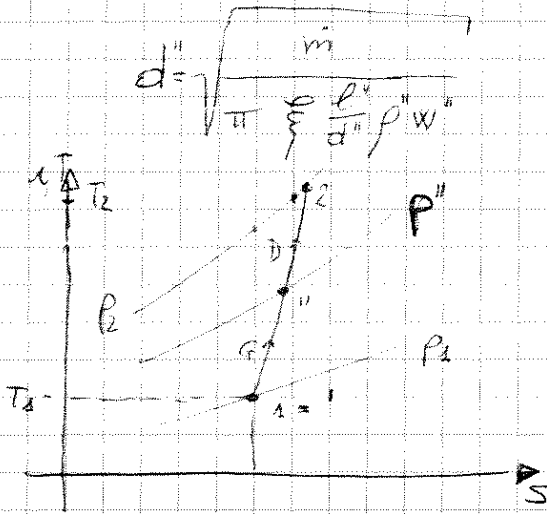
$$u'' = \sqrt{L_i} = 367,3 \text{ m/s}$$

$$w'' = u'' \tan(\alpha_2'') = 141,4 \text{ m/s}$$

$$c'' = \sqrt{u''^2 + w''^2} = 412,5 \text{ m/s}$$

$$\dot{m} = \rho'' \int (\pi d'' l'') w'' = \rho'' \int \frac{l''}{d''} \pi d'' w''$$

componente di  
rispetto delle vortici



$$\rho'' = \frac{p''}{RT''}$$

Scriviamo il 1° principio del diffusore

$$0 = q(T_2 - T'') + \frac{c_2^2 - c''^2}{2}$$

$$\frac{c_1''}{c_1'} \ll \frac{w_1''}{w_1'}$$

de  $c_2 \ll c''$  perché è la velocità del fluido  
nel condotto ad esso compresso

$$T'' = \frac{c_p T_2 - \frac{c''^2}{2}}{c_p} = 79,9 \text{ °C} \quad c_p = 1004 \frac{\text{J}}{\text{kgK}}$$

$$p'' = p' \left( \frac{T''}{T_1} \right)^{\frac{\gamma}{\gamma-1}} = 1,71 \text{ bar}$$

$$\rho'' = \frac{p''}{RT''} = 1,684 \frac{\text{kg}}{\text{m}^3} \quad R = 287$$

↓

$$d'' = \sqrt{\frac{\dot{m}}{\rho'' \pi \int \frac{l''}{d''} w''}} = 0,061 \text{ m} \Rightarrow l'' = \frac{l'}{d'} \cdot d'' = 61 \text{ mm}$$

$$n = \frac{u''}{\pi d''} = 12137 \text{ rpm}$$

Esercizio: due turbocompressori geom. simili e funzionanti

in condizione di similitudine

$$T_1 = T_2$$

Compressore I

Compressore II

$$n = 25000$$

Pos

$$30000$$

$$\dot{m}_1 = 2 \text{ kg/s}$$

?

$$P_0 = 300 \text{ Hp}$$

?

$$\beta_I = \beta_{II}$$

$$\eta_m = 1$$

$$P_{0I} = \frac{\dot{m}_I L_{cI}}{\eta_m} \Rightarrow L_{cI} = \frac{P_{0I} \eta_m}{\dot{m}_I} = \frac{300 \cdot 0,756 \cdot 1}{2} = 110,3 \frac{\text{kJ}}{\text{kg}}$$

$$T_2^I = T_1^I + \frac{L_{cI}}{c_p} = 129,1^\circ\text{C}$$

$$R = 0,059 \frac{\text{kcal}}{\text{kg}^\circ\text{K}} \Rightarrow c_p = \frac{R}{\kappa - 1} = \frac{14}{1,4 - 1} \cdot 0,069 \cdot 4,1868 = 1,011 \frac{\text{kJ}}{\text{kg}^\circ\text{K}}$$

$$\left(\frac{T_2^I}{T_1^I}\right) = (\beta_I)^{\frac{1}{\eta_{gcI}} \frac{\kappa - 1}{\kappa}} \Rightarrow \beta_I = \left(\frac{T_2^I}{T_1^I}\right)^{\eta_{gcI} \frac{\kappa}{\kappa - 1}} = 2,565$$

$$L_{cII} = c_p T_1^I \left( \beta_I^{\frac{1}{\eta_{gcI}} \frac{\kappa - 1}{\kappa}} - 1 \right)$$

Poiché i 2 compressori lavorano in similitudine hanno lo stesso rendimento. Ne segue che il lavoro è lo stesso

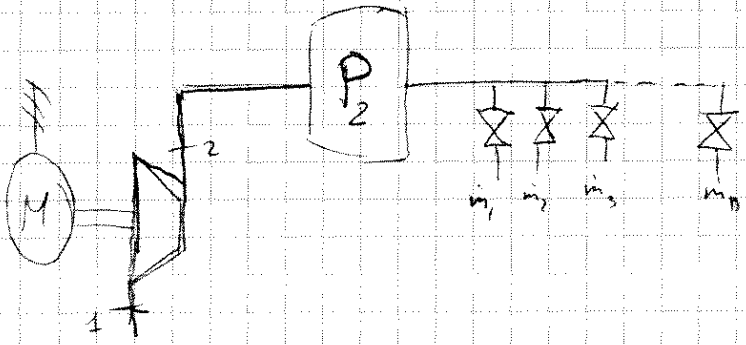
$$\frac{\dot{m}_I \sqrt{RT_1}}{\rho_I d_I^2} = \frac{\dot{m}_{II} \sqrt{RT_1}}{\rho_{II} d_{II}^2} \Rightarrow \frac{\dot{m}_I}{\dot{m}_{II}} = \left(\frac{d_I}{d_{II}}\right)^2$$

$$\frac{n_I d_I^2}{\sqrt{RT_1}} = \frac{n_{II} d_{II}^2}{\sqrt{RT_1}} \Rightarrow \frac{n_I}{n_{II}} = \frac{d_{II}^2}{d_I^2}$$

$$\frac{\dot{m}_I}{\dot{m}_{II}} = \left(\frac{n_{II}}{n_I}\right)^2 \Rightarrow \dot{m}_{II} = \dot{m}_I \left(\frac{n_I}{n_{II}}\right)^2 = 1,39 \frac{\text{kg}}{\text{s}}$$

$$P_{0II} = \frac{\dot{m}_{II} L_{cII}}{\eta_m} = 153,2 \text{ kW}$$

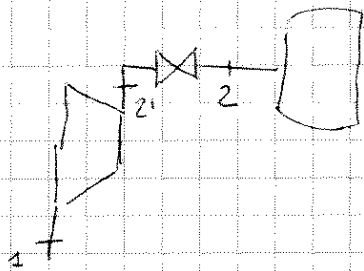
Esercizio



A portata variabile  
bisogna mantenere costante  
 $P_2$

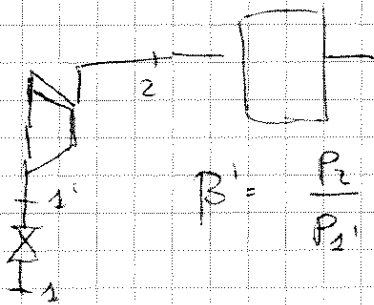
Le regolazioni da realizzare sono:

- 1) n° giri variabile ( $\beta = \text{costante}$ )
- 2) Eliminazione dello scatto



$$\beta' = \frac{P_2'}{P_1} > \beta$$

- 3) Eliminazione all'aspirazione



$$\beta' = \frac{P_2}{P_2'} < \beta$$

Esercizio

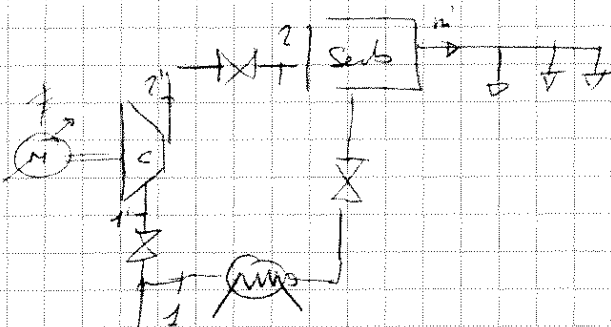
$$\frac{\eta}{\eta_0} \sqrt{\frac{T_0}{T_1}} = 1,00$$

$$\beta = 2,6 \pm$$

$$\xi \sqrt{\frac{T_0}{T_1}} \frac{P_2}{P_1} = 31,5 \frac{\text{kg}}{\text{s}}$$

$$\eta_0 = 0,9$$

$$T_1 = T_0 = 300 \text{ K} \quad P_1 = P_0 = 1 \text{ bar} \quad n_0 = n = 17000 \text{ rpm}$$



Analizzare i 3 casi  
di regolazione!

1) Ai limiti del pompaggio  $\sqrt{\frac{T_1}{T_0}} \frac{P_0}{P_1} = 27,6 \frac{\text{kg}}{\text{s}}$

Nel funzionamento iniziale

$$L_i = c_p \frac{T_1}{\eta} \left( \beta^{\frac{k-1}{k}} - 1 \right) = 108,5 \frac{\text{KJ}}{\text{kg}}$$

Perché  $T_1 = T_0$  e  $P_1 = P_0$

$$\dot{m} = \cancel{27,6} \cdot 3315 \frac{\text{kg}}{\text{s}}$$

$$P_i = \dot{m} L_i = 3,62 \text{ MW}$$

1)  $\dot{m} = 27,6 \frac{\text{kg}}{\text{s}} \quad \beta = 2,67 \quad \eta_c \approx 0,65$

$$L_i = 150,1 \frac{\text{KJ}}{\text{kg}} \quad P_i = 4,14 \text{ MW}$$

$$\frac{\eta}{\eta_0} \sqrt{\frac{T_1}{T_0}} \approx 0,865 = 11580 \text{ rpm}$$

2)  $\eta = \text{costante}$  portata costante =  $28,5 \frac{\text{kg}}{\text{s}}$

$$\dot{m} = 28,5 \text{ kg/s}$$

$$\beta' = 2,82$$

$$\eta_c \approx 0,65$$

$$P_2' = \beta' P_1 = 2,82 \text{ bar}$$

$$L_i = 158,8 \frac{\text{KJ}}{\text{kg}} \quad P_i = 4,56 \text{ MW}$$

3)  $\eta = \text{cost}$   $T_0$  rimane  $\neq$  nome  $\frac{P}{P_0}$

portata costante  $28,5 \frac{\text{kg}}{\text{s}} \quad \beta' = 2,82 \quad \eta_c = 0,65$

$$\frac{P_2'}{P_1} = \beta^k \Rightarrow P_1' = \frac{P_1}{\beta'} = P_1 \frac{\beta}{\beta'} \Rightarrow \frac{P_1'}{P_1} = \frac{P_1}{P_0} \frac{\beta}{\beta'}$$

$$\dot{m} = \frac{\text{portata costante} \cdot P_1'}{P_0} = \frac{P_1 \cdot \text{portata costante} \cdot \beta}{\underbrace{\sqrt{T_1/T_0}}_{=1} \cdot \beta'} = 27 \frac{\text{kg}}{\text{s}}$$

$$L_i = 150,8 \frac{\text{KJ}}{\text{kg}}$$

$$P_i = 4,31 \text{ MW}$$

~~4) Prof.~~

## Esercizio TURBO POMPA CENTRIFUGA MONOSTADIO CON

diffusore polidato.

Pole girante rivolte in avanti

$$\beta'' = 60^\circ$$

$$\alpha'' = 20^\circ$$

$$d'' = 0,4 \text{ m}$$

$$n = 1500 \text{ rpm}$$

$$c_1 = 8 \text{ m/s}$$

$$\eta_y = 0,75$$

perdite

{ 40% guide  
60% diffusore

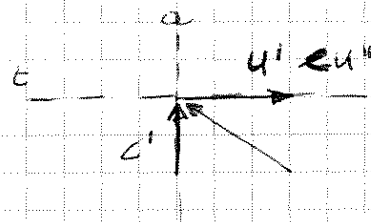
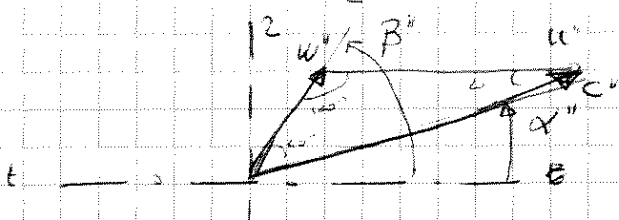
$$P_1 = 1 \text{ bar}$$

$$P_2'' = ?$$

$$c_2'' = ?$$

$$P_2' = ?$$

trascurare  $c_2 \approx 0$



$$L_i = u'' c_u''$$

$$u'' = \pi d'' n = \frac{31,4 \text{ m}}{1} \text{ s}$$

$$\frac{u''}{\sin 40^\circ} = \frac{c''}{\sin 120^\circ} \Rightarrow c'' = u'' \frac{\sin 120^\circ}{\sin 40^\circ} = 47,3 \frac{\text{m}}{\text{s}}$$

$$L_i = u'' c_u'' = u'' c'' \cos \alpha'' = 1175 \frac{\text{kJ}}{\text{kg}}$$

$$L_i = \frac{p'' - p'}{\rho} + \frac{c''^2 - c_1^2}{2} + L_w$$

$$L_i = \int_1'' \kappa dp + \Delta E_c |_1'' + L_w |_1''$$

$$\eta_y = \frac{L_i - L_w}{L_i} = 0,75 \quad L_w = (1 - \eta_y) L_i = 0,31 \frac{\text{kJ}}{\text{kg}}$$

$$L_w g = 0,115 \frac{\text{kJ}}{\text{kg}}$$

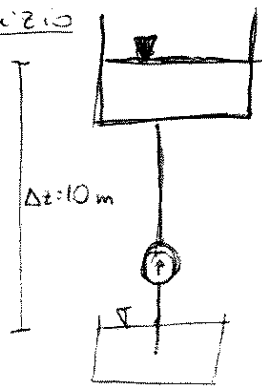
$$L_w d = 187,4 \frac{\text{J}}{\text{kg}}$$

$$\frac{p''}{\rho} = L_i - \frac{c''^2 - c_1^2}{2} - L_w g + \frac{p'}{\rho} = 358 \frac{\text{J}}{\text{kg}}$$

$$p'' = \frac{p'}{\rho} \cdot \rho = 3,58 \text{ bar}$$

Esercizio

turbo pompa centrifuga



$$Y(Q = 17,5 \text{ l/s}) = 0,5 \text{ m}$$

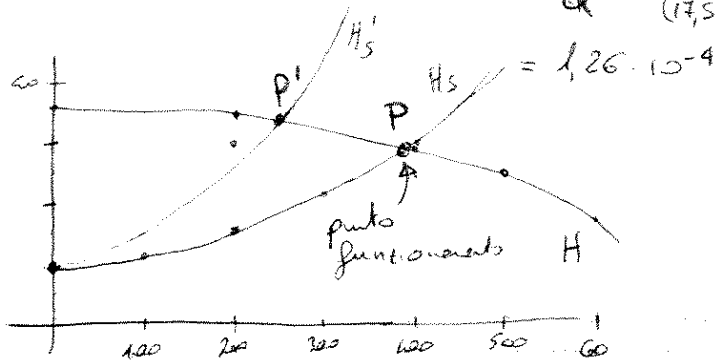
$$\eta_0 = 0,92 \quad \eta_v \cong 1 \quad n = 1500 \text{ rpm}$$

$$P_a = \frac{\rho g H Q}{\eta_0 \eta_y \eta_v}$$

$$H_s = \Delta z + Y = \Delta z + KQ^2$$

$$Y(Q = 17,5 \text{ l/s}) = KQ^2 = 0,5 = \text{?} \quad K = \frac{Y}{Q^2} = \frac{0,5}{(17,5 \cdot 3,6)^2} = 1,26 \cdot 10^{-4}$$

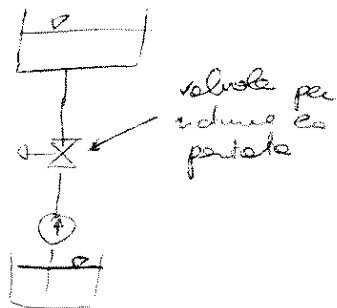
Q (m <sup>3</sup> /h)	H <sub>s</sub> (m)
0	10
100	17,26
200	15,06
300	21,34
400	30,16
500	41,50
600	55,35



$$Q_P \cong 400 \text{ m}^3/\text{h} \quad H_p = 30 \text{ m} \quad \eta_y = 0,82$$

$$P_a = \frac{1000 \frac{\text{kg}}{\text{m}^3} \cdot 9,81 \cdot 30 \cdot 400 \frac{\text{m}^3}{\text{h}}}{0,92 \cdot 1 \cdot 0,82} = 42,3 \text{ kW}$$

Riduzione portata del 40%  $\Rightarrow Q' = 0,6 Q = 240 \text{ m}^3/\text{h}$



$$P'_a = \frac{1000 \frac{\text{kg}}{\text{m}^3} \cdot 9,81 \cdot 240 \frac{\text{m}^3}{\text{h}} \cdot 30}{0,92 \cdot 1 \cdot 0,64} = 37,7 \text{ kW}$$

Di solito  $\eta'_0 < \eta_0$  dobbiamo calcolare le perdite

$$P_{\text{ass}} = P_i + P_0$$

$$P'_{\text{ass}} = P'_i + P_0$$

$$\frac{1}{\eta'_0} = \frac{1}{\eta_0} + \frac{P_0}{P_i}$$



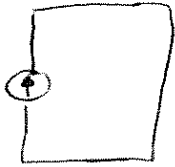
Esercizio (3)

Pompe centrifuga mono-stadio

$n = 1500 \text{ rpm}$      $Q = 200 \text{ l/s}$      $\eta_P = 0,75$

$Y = 40 \text{ m}$      $Y \propto \omega^2$      $\Delta z = 0$

$Q' = 150 \text{ l/s}$      $\frac{Q'}{Q} = \frac{n'}{n}$      $\frac{H'}{H} = \left(\frac{n'}{n}\right)^2$



$$P'_0 = \frac{\rho g H' Q'}{\eta_v \eta_m \eta_g} =$$

Nota:  $\eta_g' = \eta_g$  perché la curva di funzionamento è un parabola passante per l'origine con la curva isoefficienza (perdi solo su circuito chiuso)

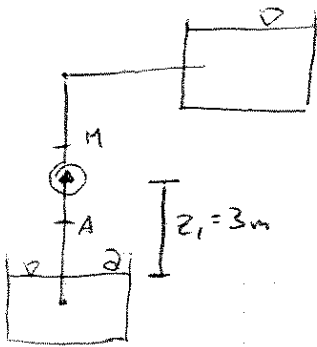
Esercizio (1)

$n = 800 \text{ rpm}$      $Q = 80 \text{ l/s}$      $Y \propto \omega^2$

$Y(Q=100 \text{ l/s}) = 0,6 \text{ m/m}$

$p_{amb} = 1 \text{ bar}$

$T_{amb} = 20^\circ\text{C}$



$Y_A|_{Q=100 \text{ l/s}} = 0,6 \cdot z_1 = 1,8 \text{ m}$

$NP_{SH} > NP_{SH_{min}}$

$$\frac{p_{amb} - p_v(T_a)}{\rho g} - z_1 - Y_a < NP_{SH_{min}}$$

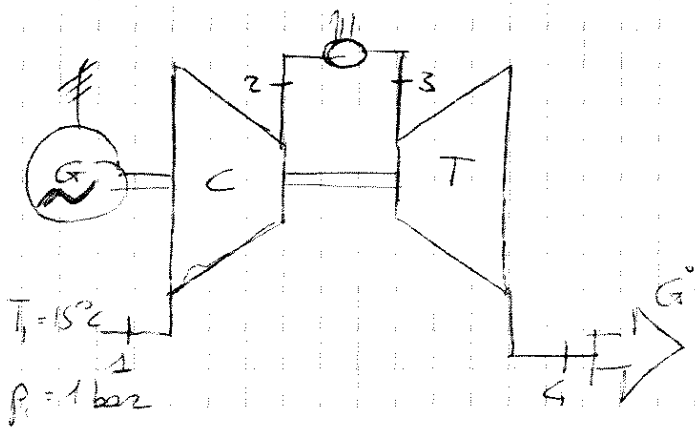
$$\frac{(1 - 234 \cdot 10^{-3})}{1000 \frac{\text{kg}}{\text{m}^3} \cdot 9,81 \frac{\text{m}^2}{\text{s}^2}} - 3 \text{ m} - Y_a|_{Q=80 \text{ l/s}} \approx 5,50 \text{ m}$$

$$Y_a|_{Q=80 \text{ l/s}} = Y_a|_{Q=100 \text{ l/s}} \cdot \frac{80^2}{100^2} =$$

$NP_{SH_{min}}(Q=80 \text{ l/s}, n=800) = 2 \text{ m}$

$n' = 1750 \text{ rpm}$

Esercizio: ciclo ~~gas~~ combusto INEL Livorno Feneno



$$\beta_c = 14 = \beta_t \Rightarrow \eta_{\pi} = 1$$

$$\dot{m}_g = 1605 \text{ t/h} = 448,9 \frac{\text{kg}}{\text{s}}$$

$$T_u = 505^\circ\text{C}$$

$$K_g = 1,333$$

$$c_{pg} = 1,147 \frac{\text{kJ}}{\text{kg K}}$$

$$\eta_{gt} = 0,865$$

$$\eta_{gt} = 0,865$$

$$H_i = 46700 \text{ kJ/kg}$$

$$\eta_b = 0,995$$

$$\eta_{me} = 0,985$$

$$K_d = 1,4 \quad c_{pe} = 1,005 \frac{\text{kJ}}{\text{kg K}}$$

Compressore

$$T_2 = T_1 \beta^{\frac{1}{\eta_{gc}}} \frac{K_g - 1}{K_g} = 415,8^\circ\text{C}$$

Turbina combusta

$$\textcircled{1} \eta_b H_i = (1 + \alpha) c_{pg} (T_3 - T_2) \quad \text{form. appross}$$

$$\textcircled{2} \eta_b H_i + \alpha c_{pe} (T_c - T_0) = (\alpha + 1) c_{pg} (T_3 - T_0) \quad \text{formule esatte}$$

$$\alpha = \frac{\eta_b H_i - c_{pg} (T_3 - T_0)}{c_{pg} (T_3 - T_0) - c_{pe} (T_c - T_0)} = 53,6$$

$$\alpha = \frac{\eta_b H_i - c_{pg} (T_3 - T_2)}{c_{pg} (T_3 - T_0)} =$$

Turbina:  $T_3 = T_u \beta^{\eta_{gt}} \frac{K_g - 1}{K_g} = 1103,8^\circ\text{C}$

$$\dot{m}_g = \frac{\dot{m}_g}{\alpha + 1} = 8,16 \text{ kg/s}$$

$$\dot{m}_c = \dot{m}_g - \dot{m}_{ig} = 437,6 \text{ kg/s}$$

$$P_{e, TG} = \eta_{me} [P_{iT} - P_{iC}] = \eta_{me} [\dot{m}_g c_{pg} (T_3 - T_u) - \dot{m}_c (T_2 - T_u)] = 127870 \text{ kW}$$

$$\dot{Q} = \dot{m}_g c_{pg} (T_u - T_o) = 245440 \text{ kW}$$

$$\eta_{eTC} = \frac{P_{eTC}}{\dot{Q}} = 0,336$$

$$\dot{F} = \dot{m}_g \cdot H_i = 381,072 \text{ MW}$$

Generatore di vapore a recupero a 2 livelli di presso

Dati: AP: 52,9 bar

BP: 6,28 bar

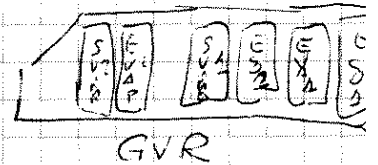
$$\Delta T_{AP_2} = T_4 - T_{e2} = 15^\circ\text{C} \Rightarrow T_{e2} = T_4 - \Delta T_{AP_2} = 630^\circ\text{C}$$

$$\Delta T_{TP_2} = T_{e2} - T_{c2} = 25^\circ\text{C}$$

$$T_{b2} = T_{c2} + \Delta T_{TP} = 257,5^\circ\text{C}$$

$$T_{c2} = 267,5^\circ\text{C}$$

p. TO	P (bar)	T (°C)	i (kJ/kg)
e <sub>2</sub>	52,9	630	3407
d <sub>2</sub>	"	267,5	2731
c <sub>2</sub>	"	267,5	1172
e <sub>1</sub>	6,28	273	3064
d <sub>1</sub>	"	160,7	2759
c <sub>1</sub>	"	"	1758
b <sub>1</sub>	6,28	"	220
a <sub>1</sub>	0,14 bar	52,6	220 ↓



$$\dot{m}_{v2} = \frac{\dot{m}_g c_{pg} (T_u - T_{e2})}{i_{e2} - i_{c2}} = 43,6 \text{ kg/s}$$

$$\Delta T_{AP_1} = T_{b2} - T_{e1} = 13,5^\circ\text{C}$$

$$\Delta T_{TP} = T_{b1} - T_{c1} = 25^\circ\text{C}$$

$$T_{b1} = 160,7 + 25 = 185,7^\circ\text{C}$$

$$\dot{m}_g c_{pg} (T_{e2} - T_{b1}) = \dot{m}_{v1} (i_{e1} - i_{c1}) + \dot{m}_{v2} (i_{e1} - i_{b2})$$

$$\dot{m}_{v1} = 13,2 \text{ kg/s}$$

Nell'ecosistema  $\dot{m}_{v1} + \dot{m}_{v2}$

$$\dot{m}_g c_{pg} (T_{b1} - T_s) = (\dot{m}_{v1} + \dot{m}_{v2}) (i_{c1} - i_{b1})$$

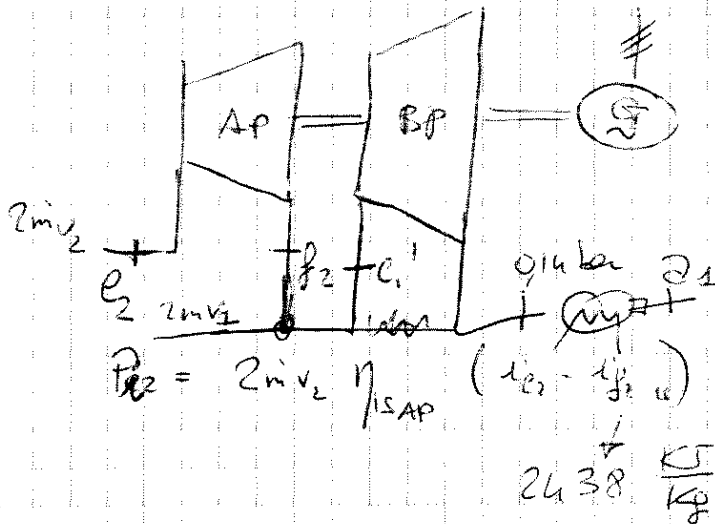
$$T_s = 120,3^\circ\text{C}$$

$$\dot{Q}_S = \dot{m}_g c_{pg} (T_S - T_0) = 53,803 \text{ MW} \quad \text{potencia de potencia} \\ \text{antes repouca}$$

$$\dot{Q}_V = \dot{Q} - \dot{Q}_S = 191,597 \text{ MW}$$

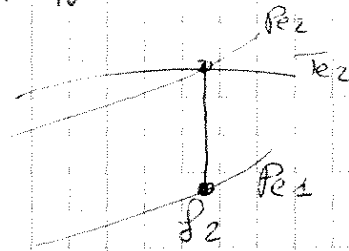
$$\epsilon_{EVR} = \frac{\dot{Q}_V}{\dot{Q}} = \frac{191,597}{245,440} = 0,781$$

Impianto a repouca (conf. 2.1)



$$\eta_{is, AP} = 0,8 = \eta_{is, BP}$$

$$\eta_{me, TV} = 0,98$$



$$P_{C2} = 75506 \text{ kW}$$

$$t_{g2} = 1630 \text{ KJ/kg}$$

$$(\dot{m}_{v1} + \dot{m}_{v2}) t_{e3}' = \dot{m}_{v1} t_{e3} + \dot{m}_{v2} t_{g2} = \dot{m}_{v1}' t_{e1}' = 2710 \frac{KJ}{kg}$$

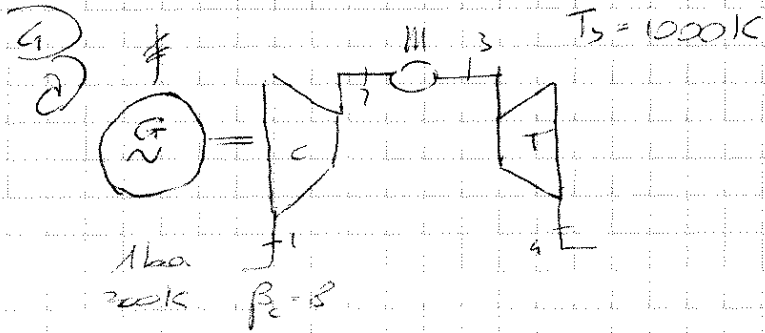
$$t_{g1, is} = 2105 \text{ KJ/kg}$$

$$P_{i, 1} = 2(\dot{m}_{v1} + \dot{m}_{v2}) \eta_{is, BP} (t_{g1}' - t_{g2, is}) = 59,927 \text{ MW}$$

$$P_{E, TV} = \eta_{me, TV} (P_{i, 1} + P_{i, 2}) = 132232 \text{ kW}$$

$$\eta_{sc} = \frac{2 \cdot P_{E, TV} + P_{E, TV}}{2 \cdot H \cdot \dot{m}_g} = 0,50$$

Exercice: turbine a gas



$\eta_{gc} = 0,85$   
 $\eta_{gt} = 0,88$   
 $\eta_b = 0,93$      $\eta_o = 0,97$   
 $H_c = 42,7 \frac{MJ}{kg}$   
 $\kappa = \kappa' = 1,4$

$P_u = 10 \text{ MW}$        $\dot{m}_a = ?$

$P_u = \eta_o P_c = \eta_o \dot{m}_a L_c = \eta_o \dot{m}_a \left[ \frac{\alpha+1}{\alpha} L_t + L_c \right]$

$L_c = c_p T_2 \left( \beta_c^{\frac{1}{\kappa} \frac{\kappa-1}{\kappa}} \right) = 1,0045 \cdot 300 \cdot 8^{\frac{1}{0,85} \frac{1,4-1}{1,4}} = 304,9 \text{ kJ/kg}$

$T_2 = T_1 + \frac{L_c}{c_p} = 603,5 \text{ K}$

$\alpha = \eta_b H_c = (1+\alpha) c_p (T_3 - T_2) \rightarrow \alpha = 99,7$

$L_t = \eta_{gt} c_p T_3 \left( 1 - \beta_c^{-\frac{\kappa-1}{\kappa}} \right) = 396 \frac{\text{kJ}}{\text{kg}}$

$T_4 = T_3 - \frac{L_t}{c_p} = 605,8 \text{ K}$

$L_i = \frac{\alpha+1}{\alpha} L_t = L_c = 95,1 \frac{\text{kJ}}{\text{kg}}$

$\dot{m}_a = \frac{P_u}{\eta_o L_i} = 108,4 \frac{\text{kg}}{\text{s}}$

$\eta_g = \frac{P_u}{\dot{m}_b H_c} = \eta_o \eta_b \frac{L_c}{Q_1}$

$Q_1 = c_p (T_3 - T_2) = 388,3 \frac{\text{kJ}}{\text{kg}}$

$\eta_g = 0,97 \cdot 0,93 \frac{95,1}{388,3} = 0,215$

b)  $T_3' = 300 \text{ K}$

$L_t' = \eta_{gt} c_p T_3' \left( 1 - \beta_c^{-\frac{\kappa-1}{\kappa}} \right) = 250,6 \frac{\text{kJ}}{\text{kg}}$

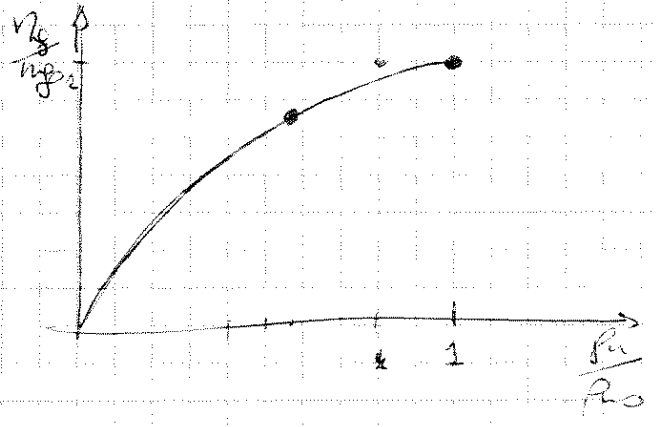
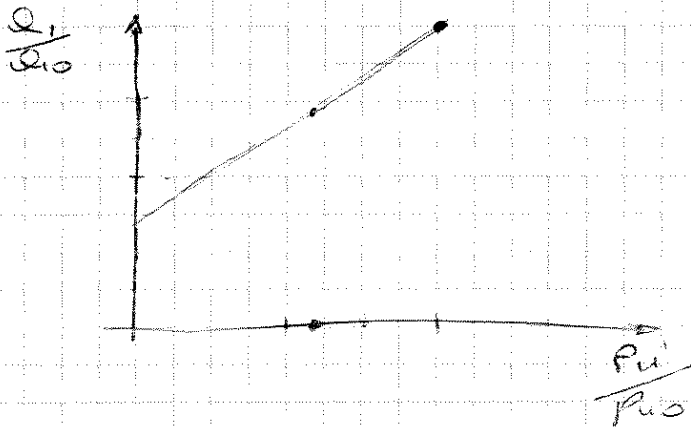
$\alpha' = \frac{\eta_b H_c}{c_p (T_3' - T_2)} = 132,9$

$Q_1' = c_p (T_3' - T_2) = 297,8 \quad \Rightarrow \quad \frac{Q_1'}{Q_1} = 0,768$

$\frac{\dot{m}_b'}{\dot{m}_b} = \frac{Q_1'}{Q_1} = 0,768$

$$L_i' = \frac{\alpha+1}{\alpha} L_r' + L_c = 54,7 \frac{\text{KGS}}{\text{kg}}$$

$$\eta_g' = \eta_b \eta_b \frac{L_i'}{a_i'} = 0,164$$



$$\Delta Q_i \propto \Delta T_3$$

$$\Delta P_u \propto \Delta L_i' \propto \Delta L_r \propto \Delta T_3$$

Può si essere di ridurre la potenza abbassando  $T_3$ . Oggi:  
si dice di ridurre ma senza far diminuire  $T_3$ . (Caricare  
ad  $\alpha$  costante)

### Esercitazione rotor alternativi.

1)

$$p_{me} = \frac{P_u}{\frac{1}{10} V \frac{\eta}{m}} = 13,06 \text{ bar}$$

$$n = 2000 \text{ rpm}$$

$$V = \frac{1}{10} V \frac{\eta}{m} = \frac{\pi d^2}{4} c \quad d = 2c$$

$$c^3 = \frac{4}{\pi \eta} \frac{1}{10} V \frac{\eta}{m} = c \quad c = \sqrt[3]{\frac{1}{10} V \frac{\eta}{m}} = 45,7 \text{ mm}$$

$$u = 27,4 \text{ m/s}$$

3)

$$\frac{P_u}{1/10} = 2000 \text{ kW}$$

$$V = \frac{\pi d^2}{4} c$$

$$d = 640 \text{ mm}$$

$$u = 10 \text{ m/s}$$

$$u = 2000 \text{ rpm} \Rightarrow c = \frac{u}{2n} = 300 \text{ mm}$$

$$V = 0,28 \text{ m}^3$$

$$p_{me} = \frac{P_u}{\frac{1}{10} V \frac{\eta}{m}} = 24,35 \text{ bar}$$

5)  $V_D = 1600 \text{ cm}^3$

$V_D = 1300 \text{ cm}^3$

$n = 3000$

$C_d = 38 \text{ Nm}$

$C_o = 120 \text{ Nm}$

$q_b = 260 \text{ g/kWh}$

$q_{bo} = 270 \text{ g/kWh}$

$\dot{m}_{aD} = 160 \text{ kg/h}$

$\dot{m}_{aO} = 160 \text{ kg/h}$

$p_a = 1 \text{ bar}$      $T_a = 20^\circ \text{C}$

$p_{me} = ?$

$\alpha = ?$

~~$\lambda = ?$~~  coeff. di rendimento

$p_{me} = \frac{C_u \cdot 2\pi \cdot n}{V_{tot}} = \Delta$

$p_{meD} = 6,48 \text{ bar}$

$p_{meO} = 3,42 \text{ bar}$

$q_b = \frac{\dot{m}_b}{P_u}$

$P_u = C_u \cdot 2\pi \cdot n = \Delta$

$P_{uD} = 3019 \text{ kW}$

$P_{uO} = 3770 \text{ kW}$

$\Downarrow$   
 $\dot{m}_b = P_u \cdot q_b \Rightarrow$

$\dot{m}_{bD} = 7385 \frac{\text{kg}}{\text{h}}$

$\dot{m}_{bO} = 10179 \frac{\text{kg}}{\text{h}}$

$\alpha = \frac{\dot{m}_a}{\dot{m}_b} \Rightarrow$

$\alpha_D = 21,65$

$\alpha_O = 13,75$

$\lambda_v = \frac{\dot{m}_a}{P_{tot} \cdot \frac{n}{m} \cdot V} \rightarrow$

$\lambda_{vD} = 0,79$

$\lambda_{vO} = 0,82$

$f = \frac{p_a}{RT}$

b)  $q_b' = q_b$

$q_b' = 1,1 q_b$

$\alpha' = \alpha_{ST} = 14,6$

$n' = n$

$p_{me}' = \frac{C_u' \cdot 2\pi \cdot n'}{V_{tot} \cdot \frac{n}{m}} =$

$\frac{C_u' \cdot 2\pi \cdot n'}{V_{tot} \cdot V} = \Delta$

$p_{meD}' = 8,24 \text{ bar}$

$p_{meO}' = 4,71 \text{ bar}$

$\dot{m}_b = P_u' \cdot q_b'$

$P_u = C_u' \cdot 2\pi \cdot n' = \Delta$

$P_{uD}' = 15,385 \text{ kW}$

$P_{uO}' = 16,85 \text{ kW}$

$\Downarrow$   
 $\dot{m}_{bD}' \approx 3,69$

$\dot{m}_{bO}' \approx 0,55 \dot{m}_b$

$\alpha' = \frac{\dot{m}_a'}{\dot{m}_b'} \Rightarrow$

$\alpha'_D = \frac{\dot{m}_a}{0,55 \dot{m}_b} = 63,2 = ? \alpha$

$\alpha'_O = \alpha_{ST} = 14,6 \Rightarrow \dot{m}_a' = 81,7 \frac{\text{kg}}{\text{h}}$

# Turbine idrauliche

① modello  
 $n^* = 300 \text{ giri/min}$   
 $P_u^* = 35 \text{ kW}$

$\eta_{T_{\max}}^* = 0,87$      $Q_s = 1,1 \text{ m}^3/\text{s}$   
 $D^* = 0,8 \text{ m}$

reale     $Q_e = 100 \text{ m}^3/\text{s}$      $P_u = 35 \text{ kW}$

$$n_c = \frac{n \sqrt{P_u}}{H^{5/4}} = n_c^{\text{modello}} = \frac{n^* \sqrt{P_u^*}}{H^{5/4}}$$

$$P_u^* = \eta_{T_{\max}}^* \rho g H^* Q_s^* = Q_s^* = \frac{Q^*}{D^{*2} \sqrt{H^*}}$$

$$P_u^* = \eta_{T_{\max}}^* \rho g H^{*3/2} D^{*2} Q_s^* \rightarrow H^* = 73,7 \text{ m}$$

$n_c = 476,5 \text{ giri/min}$  (NOTA  $P_u$  in realtà  $\rightarrow 1 \text{ kW} \approx 750 \text{ cv}$ )

$$P_u = \eta_{T_{\max}}^* \rho g H Q = D \quad H = 41 \text{ m}$$

$$Q_s = Q_s^* = \frac{Q}{D^2 \sqrt{H}} \Rightarrow D = \sqrt{\frac{Q_s \sqrt{H}}{Q}} = 3,768 \text{ m}$$

$$n_{\text{reale}} = \frac{n \sqrt{P_u}}{H^{5/4}} \Rightarrow n = \frac{n_c H^{5/4}}{\sqrt{P_u}} = 276,7 \text{ rpm}$$

$$n_s = n_s^* = \frac{n^* D^*}{\sqrt{H^*}} = 133,4 \text{ rpm}$$

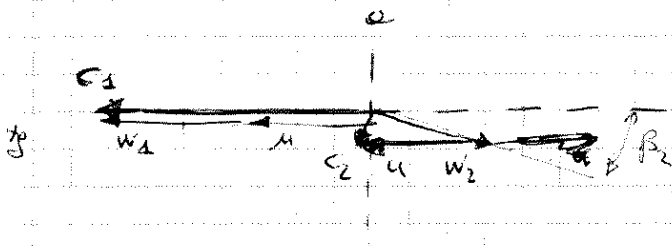
②  $Q = 8,88 \text{ m}^3/\text{s}$      $H = 721 \text{ m}$      $m = 300 \text{ rpm}$      $D = 3,54 \text{ m}$   
 $P_u = 55175 \text{ kW}$      $\eta_m = 0,97$      $\psi = 0,94$      $\beta_1 = 10^\circ$

$$P_u = \eta_T \rho g H Q = \eta_m \underbrace{\eta_v \eta_g}_{L_i} g H \rho Q$$

$$L_i = \frac{P_u}{\eta_m \rho Q} = 6405,6 \frac{\text{J}}{\text{kg}} \approx 6,4 \frac{\text{kJ}}{\text{kg}}$$

$$L_i = u (c_{u1} - c_{u2})$$





$$c_{u1} = c_1$$

$$w_2 = \psi w_{215}$$

$$w_{215} = w_2$$

$$c_{u2} = m - \frac{w_2 \cos \beta_2}{\psi}$$

$$v_1 = c_1 - u$$

$$w_2 = \psi (c_1 - u)$$

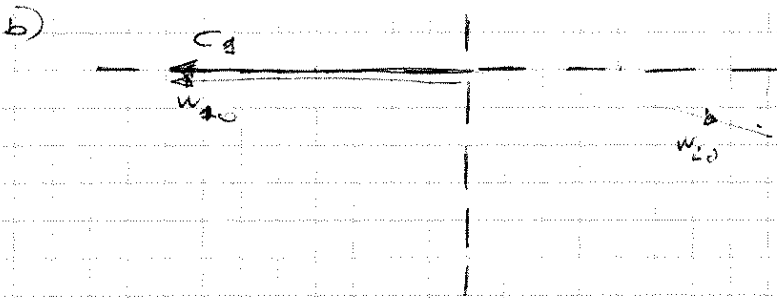
$$L_i = m (c_1 - u) (1 + \psi \cos \beta_2)$$

$$c_1 = m + \frac{L_i}{m (1 + \psi \cos \beta_2)} =$$

$$u = \pi D n = 55.61 \text{ m/s}$$

$$= 115.43 \text{ m/s}$$

$$Q = \frac{\pi d^3}{4} c_1 = Q \quad d = 0.313 \text{ m}$$



$$c_1 = w_{10}$$

$$u = 0$$

$$w_{10} = \psi w_{10}$$

$$w_{10} = c_1$$

$$c_{u20} = w_{10} = \psi c_1 \cos \beta_2$$

$$M = \frac{D}{2} (c_{u1} - c_{u20}) \rho Q =$$